

# DASH Robot: Minimalistic Design and Optimal Aerial-Terrestrial Locomotion via Contact-Implicit Control

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**Abstract**—We present a novel and minimalistic design of an aerial-terrestrial robot DASH: Ducted Aerial Spring Hopper. The goal is to enable both aerial and ground locomotion capabilities on a unified mobile robot that is mechanically-minimalistic, locomotion-versatile, and energy-efficient. We propose an organic integration of ducted fan co-axial body with a springy leg at the bottom for realization. The ducted fan module provides thrust-vectoring as the main actuation for agile flying; when it is combined with the light-weight spring leg, the robot realizes highly efficient ground hopping with energy circulation. Moreover, to realize optimal locomotion with two modes, we employ a contact-implicit model predictive control to automatically choose locomotion modes and actuation. We successfully validated the design and control of DASH through a range of tasks, including periodic hopping, aerial flight, and mode-free locomotion with autonomous mode transitions during obstacle traversal.

## I. INTRODUCTION

Mobile robots are increasingly expected to operate across diverse environments that include both structured and unstructured terrains. Legged robots [1], [2] are efficient for traversing irregular ground but struggle with discontinuities and obstacles that exceed their reach. Aerial robots can bypass such obstacles but incur high energetic cost and limited endurance [3], [4]. These complementary strengths motivate hybrid aerial-terrestrial robots that combine the efficiency of legs with the agility of flight [5]–[8]. Such systems promise to extend mobility beyond what is possible with either modality alone, enabling applications in search-and-rescue [9], inspection [10], and exploration [11].

We introduce a ducted fan-legged robot, DASH - Ducted Aerial Spring Hopper, that is capable of both hopping and flying as illustrated in Fig. 1. The design integrates a spring-loaded leg for terrestrial hopping [6], [8] with ducted fans [4] for vertical thrust and assisted aerial maneuvers. This combination enables the robot to traverse flat terrain, negotiate cluttered environments, and cross large gaps by seamlessly transitioning between ground and air modes. Compared to other state-of-the-art aerial-terrestrial robots that have open-propeller systems, such as PogoX [6] and PogoDrone [12], DASH with the ducted fan [13] offer high thrust-density, compactness, enhanced safety, and potential capabilities to perform efficient physical interactions with the environment. Additionally, the duct acts as a guard against debris or foreign objects entering the blades, enhancing durability.

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The hardware design and control software will be open source after the double-blind review process.

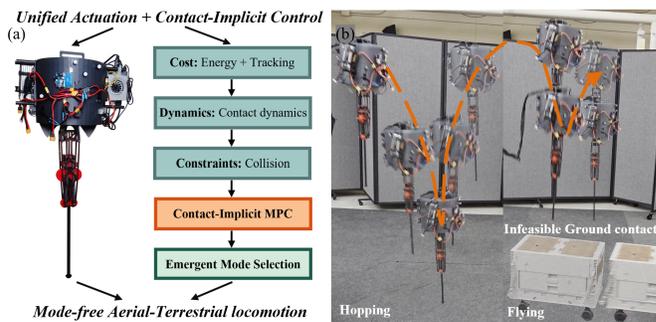


Fig. 1. (a) The hybrid robot DASH, equipped with unified actuation, addresses the challenge of mode-free aerial-terrestrial locomotion through Contact-Implicit Control. (b) DASH autonomously transitions between hopping and flight in response to environmental constraints, balancing feasibility and optimality.

Despite the promise of hybrid platforms, locomotion control remains a central challenge. Robot dynamics inherently couples continuous evolution with discrete contact events such as stance, liftoff, and landing. Conventional approaches have mainly focused on the control realization of individual modes, biased toward terrestrial locomotion [6]. The exploration of multi-modal locomotion typically relies on predefined mode schedules [12], [14], [15], which restricts optimality since all possible contact sequences must be anticipated in advance. Another common approach uses layered designs [16]–[19] that combines sampling based planning and low-level control. However, lower-dimensional models are used in the planning, which restricts the optimal solution from being found or dynamic constraints from being respected in dynamic maneuvers.

To address these challenges, we adopt the Contact-Implicit Controls (CIC) [20]–[24] in the legged robotics and manipulation community to solve the planning and control problem on multi-domain locomotion. The CIC eliminates the need for manually specified contact sequences that are also the choices of flying and hopping on aerial-terrestrial robots. Therefore, CIC allows the robot to automatically determine when and where contacts occur. Locomotion behaviors thus emerge directly from solving the control problem rather than from handcrafted logic. For DASH, this capability is critical: it enables seamless transitions between hopping and flight without explicit mode switching, under a unified cost design shaped around energy circulation.

We demonstrate that together, these contributions highlight how contact-implicit control, coupled with ducted-fan robot design, expands the mobility and versatility of hybrid aerial-terrestrial robots in environments requiring seamless

hybrid operation.

## II. RELATED WORK

**Aerial-Terrestrial Robots:** The early literature on aerial-terrestrial robots have focusing on enabling wheeled locomotion [19], [25], [26] on flying vehicles. Recent studies have also been exploring the potentials of enabling legged locomotion with flying. PogoX [6], and Hopcopter [8] demonstrated hopping–flying systems that exploit low thrust-to-weight ratio and elastic energy storage to improve terrestrial efficiency and durability. These ideas were extended to insect-scale platforms [5]. Other embodiments include quadrotor-assisted bipeds [27] and flying–crawling quadrotors [16], showing the diversity of system-level innovations at a preliminary stage in hybrid aerial-terrestrial locomotion.

**Ducted Fan Robot:** Ducted fans on aerial vehicles [28] and ducted fan vehicles [4], [13], [29] have been well studied and used on aerial vehicles as they can offer advantages over open rotors, including higher thrust efficiency, reduced tip losses, and safer operation in cluttered environments. Yet, they have been rarely applied to robotics with a few expectations, such as enabling flying of DRAGON [30] and bipedal robots [31]. The control of the ducted fan vehicles typically require several control surfaces to direct the thrust direction for attitude control. Similar ideas have been widely explored on aerial vehicles [3] to enable agile flying maneuvers. With this being said, ducted fans have not yet been applied to hybrid mobile robots despite their potential benefits over open propellers.

**Control of Hybrid Robots:** On the control side, by and large, the decoupling approaches are commonly taken that the flying and ground controllers are separately implemented. Flying behaviors are typically implemented via linear controls [8], [12], while ground locomotion, such as hopping, is typically implemented via step-level stepping controllers [6] on top of height controllers. Besides, traditional model predictive control (MPC) have been applied to wheeled aerial robot [18], and state-of-the-art data-driven predictive control [14] has demonstrate high-accuracy locomotion of flying and hopping. At the planning level, self-supervised cost-of-transport estimation [17] offers a scalable way to optimize multimodal trajectories, complementing earlier work on energy-efficient hybrid motion planning using sampling-based methods [19]. Yet, to the best of our knowledge, optimal control of hybrid locomotion across multiple domains with different dynamics remains underexplored.

**Contact Implicit Control:** Planning and control around contact has been an essential topic in legged locomotion and robotic manipulation. Common approaches often utilize soft-contact modes [32], alternative smoothing of vector fields [33], or complementarity conditions of contact [34] to make contact-implicit in the optimization of planning and control. Complementarity-based formulations of rigid-body contact dynamics [35], [36] have recently been widely adopted in Contact-Implicit MPC (CI-MPC) frameworks [20]–[22] to address legged locomotion and manipulation problems. However, the potential of CI-MPC for planning across diverse

locomotion modes in hybrid mobile robots remains largely unexplored, which constitutes a central focus of this work.

## III. ROBOT DESIGN

DASH is a compact hybrid robot that integrates thrust-assisted aerial locomotion with spring-loaded terrestrial hopping. Its architecture combines a ducted fan for lift and thrust vectoring with a pogo-stick-inspired leg for ground interaction. To support these functions, the system is built around lightweight structural components, high-power actuation, and onboard computation for real-time control.

### A. Mechanical Design

DASH combines a ducted fan propulsion unit with a pogo-stick spring leg to enable both aerial and terrestrial locomotion. The central body houses the fan assembly, enclosed by a 3D-printed duct using PLA-Aero for safety, aerodynamic efficiency, and structural support. The fan to is composed of two motorized co-axial propellers that are mounted in the duct via light-weight carbon-fiber tubes and 3D-printed PLA-CF materials. Attached to the bottom of the duct are 3D-printed vanes that provide thrust vectoring for stabilization and maneuverability. The ducted body is designed to have a radius of 10 inches to have sufficient moment control via thrust vector.

The leg is designed as a pogo-stick spring mechanism, enabling DASH to store and release elastic energy for efficient hopping maneuvers. The elastic tubing spring also reduces the thrust required for fast takeoffs when flying from hopping. A pair of spaced pulley constrains the leg’s vertical motion, providing smooth translation while minimizing slack compared to a single linear bearing. This novel integration of the ducted fan with the spring leg allows DASH to combine energy-efficient terrestrial motion with thrust-assisted aerial capabilities.

The total structural weight of the 3D-printed duct body and vanes is 719 g, balancing robustness with manufacturability. Together with propulsion and onboard electronics, the total system mass is approximately 3.27 kg, with a height of 4 feet. A schematic of the robot design is shown in Fig. 2, highlighting the integration of the pogo-stick spring leg with the ducted fan assembly.

Component	Model	Weight (g)
Battery	HRB 6S 3300mAh 60C	467
Motor	T-Motor Cine77 3610 977KV	400
Propeller	Gemfan 1050 Reinforced Nylon	34
ESC	V-Good RC 32-Bit 100A	148
Flight Controller	Holybro Pixhawk 6C Mini	39
IMU	x-io x-IMU3	50
Servo	Dynamixel XC330-T181-T	92
Computer	UP-Squared-i12	111
Duct Body	3D Print	609
Vanes	3D Print	110
Leg	Carbon Fiber	542
Wires	Harness	670
<b>Total</b>		<b>3272</b>

TABLE I. Component list with models and weights.

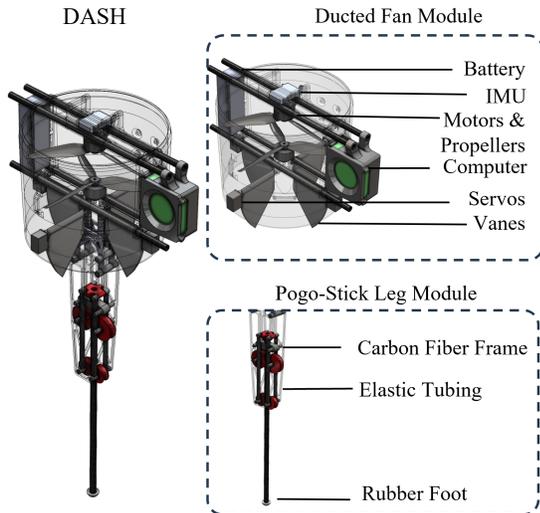


Fig. 2. Illustration of the mechanical components of DASH.

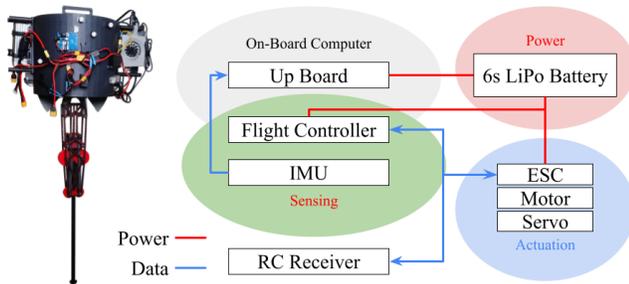


Fig. 3. Illustration of the electronics architecture of DASH.

### B. Electronics and Actuation

The electronics of DASH are designed to provide reliable high-power actuation alongside real-time computation for hybrid locomotion. The ducted fan is driven by two coaxial brushless DC motors, while thrust vectoring is realized through four servo actuators mounted on the duct. Together, these actuators generate lift and directional control for both flight and thrust-assisted terrestrial maneuvers.

Onboard computation is distributed across two systems. A PX4 flight controller manages low-level motor regulation, ensuring stable thrust generation and reliable actuator communication. A single-board computer (SBC) equipped with an Intel i7 processor executes higher-level planning and control in real time. The robot is equipped with a top-mounted IMU for orientation estimation, while translational states are obtained from a motion capture system. The system can be extended to fully onboard visual-inertial state estimation using a camera, as shown in related work [37].

All the electronics are powered by one 6S Lipo battery that supplies both propulsion and computation, with distribution designed to isolate high-current actuation from sensitive electronics. Figure. 3 summarizes the electronic architecture, while Table I provides a breakdown of the weights.

The PX4 flight controller provides off-the-shelf thrust and attitude regulation, as well as direct PWM and servo control, depending on task requirements. The SBC supplies

the computational resources for higher-level planning and control. For hybrid locomotion, DASH generates motor PWM commands and desired orientation trajectories through the CI-MPC. Both thrust and orientation are planned within the Contact-Implicit MPC framework, with thrust allocation playing a primary role in contact transitions and mode switching. The PX4 executes the PWM commands, while a quaternion-based geometric attitude controller [38] tracks the planned orientation. This architecture allows us to evaluate the proposed contact-implicit control strategy without relying on off-the-shelf flight stabilization.

## IV. ROBOT DYNAMICS

We now present the dynamic modeling of DASH. Owing to the coexistence of parameterized actuation mechanisms and hybrid locomotion behavior, we first describe the *actuation dynamics* and their key parameters, followed by the formulation of the hybrid *locomotion dynamics*.

### A. Actuation Model

DASH is actuated by the total thrust force  $\mathbf{T} \in \mathbb{R}^3$  and moment  $\boldsymbol{\tau} \in \mathbb{R}^3$  generated from the propellers and vanes, as indicated in Fig. 4. We first examine how the thrusts and moments are generated based on aerodynamics.

**Thrust Force Model:** We adopt common assumptions on flying robots that the total thrust from two propellers is:

$$\mathbf{T}_p = [0, 0, c_1\Omega_1^2 + c_2\Omega_2^2]^\top, \quad (1)$$

where  $\Omega$  denotes the rotational velocity of each propeller, and  $c_1$  and  $c_2$  are constant coefficients determined by the propellers and physical dimensions of the ducted body.

**Moment Model on Co-Axial Motors:** The two propellers also create a turning moment in the body frame z-axis:

$$\boldsymbol{\tau}_p = [0, 0, c_3(\Omega_1^2 - \Omega_2^2)]^\top, \quad (2)$$

where  $c_3$  is a parameter related to the propellers.

**Vane Force Model:** The vanes direct the airflow beneath the ducted fan. The changing of the airflow results in aerodynamic push force  $F_i$  on the vane  $i$ , where  $i = 1, 2, 3$ , or 4 is the index of the vanes. The push force  $F_i$  is modeled as:

$$F_i = \frac{1}{2}\rho ACV_o^2\alpha_i, \quad (3)$$

where  $\rho$  is the air density,  $A$  is the vane surface area,  $C$  is a constant coefficient,  $V_o$  denotes the air speed at the duct outlet, and  $\alpha$  represents the servo angle. Here, small angle approximation of  $\alpha$  is used. The air speed  $V_o^2$  is assumed proportional to the total thrust force magnitude  $T_p$ . Thus, (3) can be rewritten as:  $F_i = \frac{1}{2}\rho Ac_4 T_p \alpha_i$ , with  $c_4$  being another constant coefficient.

**Total Force:** The total thrust is obtained by summing the thrust of the propeller and the contributions of the four vane forces:  $\mathbf{T} = \mathbf{T}_p + \sum_{i=1}^4 \mathbf{F}_i$ .

**Moment:**  $\mathbf{T}_p$  and gravitational force go through the center of mass (COM). Thus the moments are only created by the Vane forces:  $\boldsymbol{\tau} = \mathbf{r} \times \mathbf{F}$ . Assuming small angles that the



**Time-stepping Dynamics:** For rigid body systems modeled by Lagrangian dynamics, the continuous time dynamics (7) can be approximated by a discrete-time system using a semi-implicit Euler integration scheme:

$$\mathbf{q}^+ - \mathbf{q} = \Delta t \mathbf{v}^+, \quad (11)$$

$$\mathbf{M}(\mathbf{q})(\mathbf{v}^+ - \mathbf{v}) = \Delta t(\mathbf{u}_r - \mathbf{h}(\mathbf{q}, \mathbf{v}) + \mathbf{J}(\mathbf{q})^\top \mathbf{f}), \quad (12)$$

where  $\mathbf{q}^+$  and  $\mathbf{v}^+$  denote the state at the next time step, and  $\Delta t$  is the discrete time interval. Instead of imposing regularized dynamics for predefined contact modes, the contact impulse  $\boldsymbol{\lambda} = \Delta t \mathbf{f}$  is obtained from a complementarity formulation of the contact constraints, enabling mode-consistent contact resolution.

**Non-penetration:** For contact at the next time step, both the signed distance between two bodies  $\phi(\mathbf{q}^+)$  and the normal contact impulse  $\lambda^n = \Delta t f^n$  during the interval must be non-negative. Moreover, normal contact impulse can only be nonzero when two bodies are in contact:  $\phi(\mathbf{q}^+) \geq 0 \perp \lambda^n \geq 0$ . A linear approximation of this condition is used for computation [40]:

$$\phi(\mathbf{q}) + \nabla \phi(\mathbf{q})(\mathbf{q}^+ - \mathbf{q}) \geq 0 \perp \lambda^n \geq 0. \quad (13)$$

**Maximum Dissipation:** The maximum dissipation principle states that tangential friction impulses act to maximize the dissipation of kinetic energy. In our single foot-contact scenario, the Coulomb friction model is approximated by a polyhedral cone, yielding the following formulation:

$$\min_{\boldsymbol{\beta}} \quad \boldsymbol{\lambda}^\top \mathbf{J}(\mathbf{q}) \mathbf{v}^+ \quad (14)$$

$$\text{s.t.} \quad \boldsymbol{\lambda} = \mathbf{n} \lambda^n + \mathbf{D} \boldsymbol{\beta}, \quad \boldsymbol{\beta} \geq \mathbf{0}, \quad (15)$$

$$\mu \lambda^n - \mathbf{e}^\top \boldsymbol{\beta} \geq 0. \quad (16)$$

where  $\mathbf{n}$  is the unit normal vector at the contact point, and  $\mathbf{D}$  positively spans the tangential contact plane with unit vectors. The vector  $\boldsymbol{\beta}$  represents the friction force coefficients along the directions defined by  $\mathbf{D}$ ,  $\mu$  is the coefficient of friction, and  $\mathbf{e}$  is a vector of ones. The Karush–Kuhn–Tucker (KKT) conditions of the associated minimization problem are expressed as the following complementarity constraints:

$$\mathbf{D}^\top \mathbf{J} \mathbf{v}^+ + \mathbf{e} \eta \geq \mathbf{0} \perp \boldsymbol{\beta} \geq \mathbf{0}, \quad (17)$$

$$\mu \lambda^n - \mathbf{e}^\top \boldsymbol{\beta} \geq 0 \perp \eta \geq 0, \quad (18)$$

where  $\eta$  denotes the Lagrange multiplier. The KKT conditions, together with the discrete-time robot dynamics (12) and the non-penetration constraint (13), give rise to a linear complementarity problem (LCP), whose solution determines the next state under contact dynamics:

$$\mathbf{x}^+ = \text{LCP}(\mathbf{x}, \mathbf{u}). \quad (19)$$

where the notation indicates that the state transition is defined through the solution of the linear complementarity problem. The LCP-based contact dynamics admit analytical gradients, which can be exploited to improve computational efficiency in optimization-based algorithms [20].

## B. Model Predictive Control

We apply model predictive control (MPC) to the dynamics of DASH formulated as a LCP (19). The MPC regulates the torso motion while planning contact interactions by balancing optimality and feasibility under the imposed costs and constraints. At each step, a nonlinear program is solved to determine an optimal sequence of a horizon of  $N$  steps. Only the first control input is executed, and the process is repeated at the next time step.

**Cost Function:** The cost  $\mathcal{J}_{\text{MPC}}$  includes trajectory tracking, control regularization, and mechanical energy tracking:

$$\mathcal{J}_{\text{MPC}} = \sum_{k=0}^T (\|\mathbf{x}_k - \mathbf{x}_{\text{ref},k}\|_{\mathbf{Q}_x}^2 + \|E(\mathbf{x}_k) - E_{\text{ref}}\|_{Q_E}^2) + \sum_{k=0}^{T-1} \|\mathbf{u}_k\|_{\mathbf{R}}^2, \quad (20)$$

where  $\mathbf{Q}_x$ ,  $Q_E$ , and  $\mathbf{R}$  denote the weights associated with each cost term,  $E(\mathbf{x}_k)$  denotes the mechanical energy of the system. The energy term enforces coherent energy circulation across contact and flight phases, supporting consistent hybrid behavior. The relative weighting between trajectory tracking and control effort determines how energy is regulated: stronger control penalization promotes passive energy exchange through terrain interaction and compliance, whereas weaker penalization allows greater active energy injection. As a result, distinct locomotion behaviors emerge from the optimization without explicit mode scheduling.

**Constraints:** The system dynamics at each time step are enforced as constraints, together with an initial state condition and control input bounds. Obstacle avoidance is enforced through a minimum-distance constraint  $\|\mathbf{p}_{\text{frame}} - \mathbf{p}_{\text{obs}}\| \geq \alpha$ , which guarantees a separation of at least  $\alpha$  between the robot frame position  $\mathbf{p}_{\text{frame}}$  and the obstacle position  $\mathbf{p}_{\text{obs}}$ .

**MPC formulation:** The MPC is concisely represented as:

$$\begin{aligned} \min_{\{\mathbf{x}, \mathbf{u}\}_{k=0}^N} \quad & \mathcal{J}_{\text{MPC}} && \text{(MPC)} \\ \text{s.t.} \quad & \mathbf{x}_0 = \tilde{\mathbf{x}}_0, && \text{(Initial State)} \\ & \mathbf{x}_{k+1} = \text{LCP}(\mathbf{x}_k, \mathbf{u}_k), && \text{(Dynamics)} \\ & \mathbf{0} \leq \text{Constr}(\mathbf{x}_k, \mathbf{u}_k), && \text{(Constraints)} \end{aligned}$$

where the bottom constraint represents all the state and input constraints.

## C. Solution Method via Differential Dynamic Programming

At each control step, we solve a finite-horizon MPC subproblem using Differential Dynamic Programming (DDP) [41]. DDP enforces dynamics implicitly by rolling out the state trajectory from the current system state and optimizes only the control sequence. The update is performed through a local quadratic approximation of the Bellman Value and Q functions, combined with a backward–forward pass. This produces a time-varying affine feedback law over the horizon that tracks the reference trajectory while respecting the LCP-based dynamics model. Since the LCP dynamics are only piecewise differentiable, the analytic gradients are relaxed to improve numerical performance [20].

**Relaxed Analytic Gradient:** Taking into account the contact mode associated with the current iterated trajectory, the

contact mode is classified into three types based on the contact complementarity constraints: no contact, clamping contact  $(\cdot)_c$ , and sliding contact  $(\cdot)_s$ . Using the specified contact modes, the original LCP problem is transformed into a set of linear velocity-level constraints:

$$\begin{bmatrix} \mathbf{v}_c^n \\ \mathbf{v}_c^t \\ \mathbf{v}_s^n \end{bmatrix} = \begin{bmatrix} \mathbf{0} \\ \mathbf{0} \\ \mathbf{0} \end{bmatrix} = \begin{bmatrix} \mathbf{J}_c^n \\ \mathbf{J}_c^t \\ \mathbf{J}_s^n \end{bmatrix} \mathbf{v}^+, \quad (21)$$

where  $\mathbf{v}$  denotes the contact-frame velocity, and  $(\cdot)^n$  and  $(\cdot)^t$  represent the normal and tangential components of the impulse, frame velocity, or Jacobian, respectively. The resulting contact force is then expressed as:

$$\begin{bmatrix} \lambda_c^n \\ \lambda_c^t \\ \lambda_s^n \end{bmatrix} = -\mathbf{A}_{cc}(\mathbf{q})^{-1} \mathbf{b}_{cc}(\mathbf{q}, \dot{\mathbf{q}}), \quad (22)$$

$$\mathbf{A}_{cc} = \begin{bmatrix} \mathbf{J}_c^n \\ \mathbf{J}_c^t \\ \mathbf{J}_s^n \end{bmatrix} \mathbf{M}^{-1} \begin{bmatrix} \mathbf{J}_c^n \\ \mathbf{J}_c^t \\ \mathbf{J}_s^n + \mathbf{J}_s^t \mu \text{sign}(\lambda_s^t) \end{bmatrix}^\top, \quad (23)$$

$$\mathbf{b}_{cc} = \begin{bmatrix} \mathbf{J}_c^n \\ \mathbf{J}_c^t \\ \mathbf{J}_s^n \end{bmatrix} \mathbf{M}^{-1} [\Delta t (\mathbf{u}_r - \mathbf{h}) + \mathbf{M}\mathbf{v}]. \quad (24)$$

Then, the gradient of the contact impulse is obtained as

$$\partial \begin{bmatrix} \lambda_c^n \\ \lambda_c^t \\ \lambda_s^n \end{bmatrix} = \mathbf{A}_{cc}^{-1} \partial \mathbf{A}_{cc} \mathbf{A}_{cc}^{-1} \mathbf{b}_{cc} - \mathbf{A}_{cc}^{-1} \partial \mathbf{b}_{cc}, \quad (25)$$

where  $\partial$  denotes the total derivative with respect to an associated parameter vector.

The analytic gradient of the contact dynamics is computed within a local region that preserves the contact mode. To provide an informative gradient that enables contact breaking, the complementarity constraints of the contact foot in the normal direction:  $\mathbf{v}^n \geq 0 \perp \lambda^n \geq 0$ , is smoothed using a relaxation variable  $\rho > 0$ :

$$\mathbf{v}^n \geq 0, \lambda^n \geq 0, \mathbf{v}^n \lambda^n = \rho. \quad (26)$$

This smoothing only takes effect in the differentiation, where velocity constraints in the normal direction are now expressed as:

$$\frac{\rho}{\lambda^n} = \mathbf{A}_{cc}^n \lambda^n + \mathbf{b}_{cc}^n, \quad (27)$$

where the implicit relation between  $\partial \lambda^n$  and  $\lambda^n$  is given by:

$$-\frac{\rho}{(\lambda^n)^2} \partial \lambda^n = \partial \mathbf{A}_{cc}^n \lambda^n + \mathbf{A}_{cc}^n \partial \lambda^n + \partial \mathbf{b}_{cc}^n. \quad (28)$$

Then the relaxed normal force gradient is given by:

$$\partial \lambda^n = -(\mathbf{A}_{cc}^n + \frac{\rho}{(\lambda^n)^2})^{-1} (\partial \mathbf{A}_{cc}^n \lambda^n + \partial \mathbf{b}_{cc}^n). \quad (29)$$

This relaxed gradient facilitates smooth transitions between contact modes when incorporated into DDP [20].

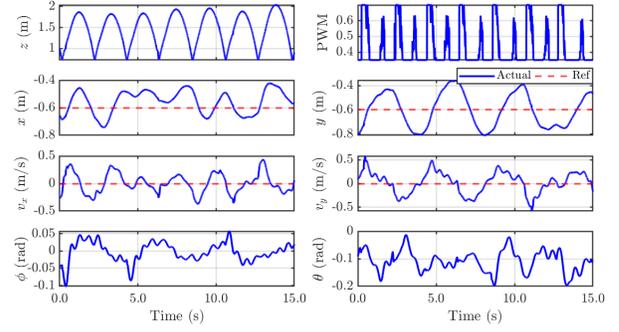


Fig. 6. Experimental results of CI-MPC controlled in-place hopping.

## VI. RESULTS

We now present the results of controlling DASH. We first present the system identification of the robot, particularly the actuation model that is rarely seen on common legged robots. Then we show the control results of the robot in simulation and hardware performing different tasks.

**System Identification:** DASH generates complex aerodynamic forces through its propellers and vanes. Rather than identifying the coefficients  $c_1$ – $c_4$  from first principles, we estimate them experimentally. To identify  $c_1$  and  $c_2$ , the robot is mounted on a single-DOF testing stand, similar to a seesaw, with a weight scale placed on the opposite side. By varying the BLDC motor speeds and measuring the resulting thrust forces, we obtain the corresponding coefficients. To identify  $c_3$  and  $c_4$ , the ducted fan assembly is mounted on a 3-DOF rotational stand. By commanding specific servo angles to adjust the vanes and generating known thrust levels, we measure the resulting moments using a force gauge, from which the coefficients are estimated. In practice, unequal speeds of the coaxial counter-rotating propellers introduce nonlinear thrust–moment coupling. We therefore enforce  $\Omega_1 = \Omega_2$ , yielding  $\tau_p = 0$  and simplifying the model without compromising achievable behaviors.

**Simulation and Software:** The robot is designed and assembled in SolidWorks, where each component is assigned its measured mass and geometric properties. This allows precise computation of the inertia and center-of-mass (COM) location. The robot is simulated using a time-stepping LCP-based contact dynamics model within a ROS2 environment. The controller and communication stack are fully implemented in C++, leveraging Crocoddyl [41] for efficient optimal control computation. The CI-MPC is configured with a 0.6 s prediction horizon and a time discretization of 0.03 s.

### A. Flying and Hopping

In the flying experiments, the robot performs hovering with global position tracking and follows a reference linear velocity of up to 0.5 m/s, constrained by the laboratory workspace as illustrated in Fig. 7.

In the hopping experiments, the robot executes continuous in place hopping while tracking global position targets with zero reference translational velocity as shown in Fig. 6. The resulting behavior maintains energy circulation while reducing control effort. Compared to flying, the optimized solution decreases the injected control effort, defined as

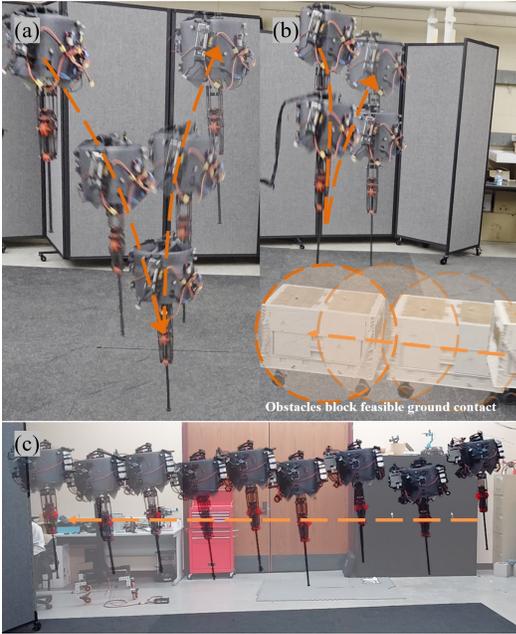


Fig. 7. Experimental demonstration of DASH. (a, b) Transition from hopping to flight planned by the CI-MPC, arising from feasibility and optimality trade-offs. (c) Flying with velocity tracking of 0.5 m/s.

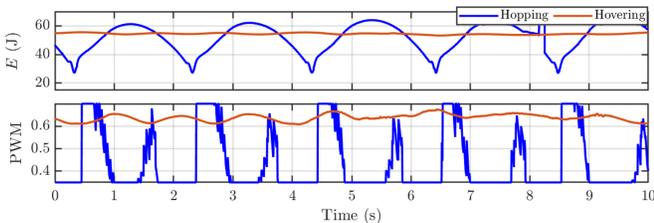


Fig. 8. Comparison of mechanical energy circulation and control effort consumption in DASH during hopping and hovering.

$\int_{t_{\text{start}}}^{t_{\text{end}}} |T(t) - T_{\text{min}}| dt$ , to 26.78% of that in the flying case, as illustrated in Fig. 8. This reduction reflects the efficient hybrid regime discovered by the CI-MPC. From Fig. 8, it can be observed that the majority of the residual energy expenditure arises from the minimal thrust command required to keep the motors active, rather than from sustained thrust injection. Unlike hovering, which requires continuous active power to counteract gravity, hopping leverages intermittent energy storage and release through contact dynamics, thereby reducing active thrust usage and improving overall efficiency.

### B. Contact-Implicit Control

We present results of mode-free motion planning, in which the CI-MPC autonomously generates hopping behavior that minimizes control effort when ground contact is feasible. Simulation results are shown in Fig. 9, where the robot traverses obstacles using mode-free locomotion. On hardware, when an obstacle appears beneath the robot, the activated collision avoidance constraints eliminate feasible contact configurations, causing the optimizer to transition the robot to flight, as illustrated in Fig. 7. This transition is not explicitly scheduled but instead emerges from feasibility considerations within the constrained optimization. From an energy perspective, when contact is feasible, the CI-MPC

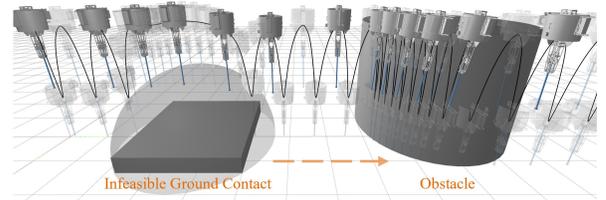


Fig. 9. Simulation results of DASH tracking a 0.3m/s velocity command, with hopping and flying emerging from feasibility-optimality trade offs.

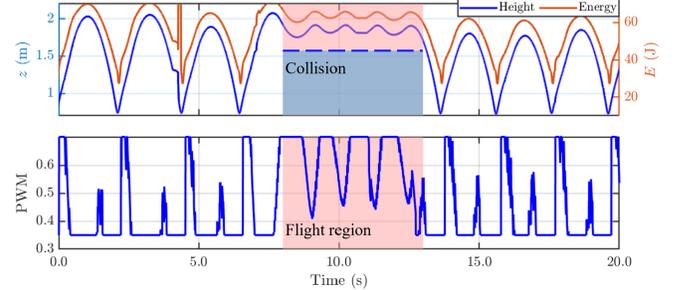


Fig. 10. Hybrid locomotion generated by the CI-MPC on hardware, where obstacle-triggered collision constraints regularize the motion toward the flight region.

exploits passive energy exchange through terrain interaction and compliant leg dynamics, resulting in hopping with reduced active thrust injection. When contact becomes infeasible due to obstacle constraints, the optimizer shifts toward flight to maintain feasibility. Once the obstacle is removed and the collision constraints become inactive, ground contact becomes feasible again, and the CI-MPC naturally returns to the hopping regime, replanning the contact sequence to recover energy-efficient behavior, as shown in Fig. 10.

## VII. CONCLUSION

To conclude, we present a controller design that enables mode free motion planning and control for the novel Ducted Aerial Spring Hopper (DASH). The proposed controller effectively integrates a contact implicit control formulation with an energy-centered cost design. A key advantage of this framework is its ability to achieve automatic locomotion mode transitions by balancing feasibility and optimality within a unified optimization problem. With optimized performance in terms of energy circulation and control effort, the resulting hopping behavior reduces the injected control effort to 26.78% of that required for hovering in hardware experiments. This improvement arises from intermittent energy storage and release through contact dynamics, which reduces the need for sustained active thrust injection.

In future work, we aim to enable DASH to operate in outdoor and natural environments, where mode-free motion planning becomes increasingly critical. We are also interested in expanding the range of locomotion behaviors beyond the current capabilities, exploring additional hybrid modes such as swimming.

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